 Shape of fuselage Cabin cross section layouts Cabin cross-section dimensions Cabin length Cockpit Fuselage main dimensions Quick method General method Fuselage weight Wing design Basic requirements Wing geometric characteristics Evaluation of Wing size Evaluation of SMC Evaluation of MAC Analytical method Graphical method Diagrammatical method Airfoils, requirements & definitions Airfoils coding High lift devices Wing aerodynamic characteristics 	
Units: 7 Weekly Hours: Theoretical: 3 Experimental: 1 Tutorial : Week Contents Introduction 1 - Design and development of aircraft - Design stages - Preliminary design department Airworthiness 2 - Definition - BCAR - - FAR - - Crash airworthiness - - Shape of fuselage - - Cabin cross section layouts - - Cabin cross section alyouts - - Cabin cross section alyouts - - Fuselage main dimensions -	
Weekly Hours: Theoretical: 3 Experimental: 1 Tutorial : Weekly Hours: Theoretical: 3 Experimental: 1 Tutorial : Weekly Contents 1 - Design and development of aircraft - Design stages - Preliminary design department 2 - Definition - BCAR - FAR - Crash airworthiness 3 - Main characteristics - Fuselage design requirements - Shape of fuselage 4 - Cabin cross section layouts - Cabin in cross-section dimensions - Cabin length - Cockpit 5 - Fuselage main dimensions - Cabin length - Cockpit 5 - Fuselage main dimensions - Quick method General method - Fuselage weight 6 - Basic requirements - Wing geometric characteristics - Evaluation of SMC - Evaluation of MAC - Analytical method - Graphical method - Graphical method - Graphical method - Graphical method - Graphical method - Analytical method - Analytical method - Analytical method - Airfoils, requirements & definitions - Airfoils coding - High lift devices - Wing acordynamic characteristics - Airfoils coding - High lift devices 9 - Lift coefficient increment due to	
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Wing aerodynamic characteristicsLift coefficient increment due to	
9 - Lift coefficient increment due to	
T.E. flaps	
Split flaps	
Single slotted flaps	
Double slotted flaps	
- Wing weight	2
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	Empennage design	
10	- Tail surfaces functions	
	- Types of surface control system	
	- Tail surface configuration	
	- Horizontal tail plane	
11	- Vertical tail plane	
	- Empennage weight	
12	Under-carriage design	
	- General requirements	
	- Runways classification	
13	- Types of undercarriage	
15		
	- Tailoring u.c. to bearing capacity of A/F	
	- LCN For single wheel	
	- LCN For two or more wheels	
	- Type, size and inflation pressure of tires	
	- Shock absorption. Leg length	
	- Ground load factor	
14	- Ground load cases	
	- Structural load cases	Dr. A. S. Al-Jaberi
	- Weight of u.c	
	Preliminary weight analysis	
15	- Weight break down	
	- Surface controls group	
	- Engine section or nacelle group	
	- Propulsion group. Engine weight	
	(dry)	
	- Airframe services and equipments	
	- Operational loads	
16	- Crew weight	
_	- Payload	
	- Fuel weight (based on flight stages)	
	- Fuel weight (based on aircraft type) (calculation by using	
	graphics)	
17	Choice of engines	
17	- Take off stages	
	- Computing of static thrust	
	- Computing minimum required thrust	
10	Center of gravity	
18	- Evaluation of aircraft center of	
	gravity	
19	- Loading and balancing diagram	
	- Wing location according to aircraft center of gravity	
•	Payload-range diagram	A PARA
20	- Limiting weight definitions	
1	- For turbo-jet aircraft	
	- For turbo-prop aircraft	10
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01	Flight and gust envelope	
21	- Load factor	
	- Load factor at steady pullout	
	- Load factor at correctly banked turn	
	- Flight envelope	
22	- Gust envelope	
	- Flight - gust envelope	
	Wing and tail loads (for flight-gust	
22	envelope)	
23	- Aircraft pitching moment of inertia	
	- Load calculation at level flight with angular acceleration	
	Span wise air and inertia load	
24	distribution	
24	- Air load distribution	
25	Schrenk method	
23	- Air load distribution	
26	Diederich method	
20	- Wing group inertia load distribution	
	- Fuselage group inertia load distribution	
		Dr. A. S. Al-Jaberi
27	Drag estimation	
	- Area drag method:	
	Wing Empennage Fuselage	
28	- Area drag method: Cockpit Undercarriage	
20	Nacelle	
	Wing/fuselage interference	
	- Empirical method for cruising stage	
	- Induced drag coefficient of wing with part-span flap	
29		
30	Structural design	
30	- Aircraft main part structural	
	design	



Subjec	ANTE 425		
v	t Number: ANTE 435		
-	et: Aircraft Propulsion Systems Tec	hnology and Design	
Units:			
Weekl	y Hours: Theoretical: 2		
	Experimental: 1		
	Tutorial : 1		
Week	Conte	ents	
	Introduction		
1	- Classification of propulsion		
	systems		
	- The thrust equation		
	Engine performance		
2	- Engine performance and A/C		
	range		
	Theory of jet propulsion	الله راکی	
3	- Turbo jet engine	J.	
4	- Turbo fan engine		
	- Turbo prop engine		
5	- Turbo shaft engine		Dr. A. S. Al-Jaberi
	- Ram jet engine		
	Design of centrifugal compressors		
6	- Impeller design		
7	- Diffuser design		
	- Volute design		
	Design of axial flow compressors		
8	- Compressor aerodynamics		
	- Repeating stage , repeating raw		
9	- Mean line design		
	- Axial variation		
	- Radial variation		
10	- Mechanical design		
	Compressors technology		
11	- Materials		
	- Manufacturing technology		
	- Balancing		
	Design of combustion system		
12	- The process (ignition, stability,		
10	length scaling, diffusers)		
13	- After burner design		
	- Flame holding		
	- Fuel injection		
1 /	Combustion chambers technology		TTT I
14	- Materials	.n. em 😤	
	- Manufacturing technology		
17	Design of axial flow turbines		
15	- Turbine aerodynamics	CHINEFELY	m I
	- Zero exit swirl, constant axial vel	ocity	All I
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16	- Mean line stage design	
	- Other design considerations	
	Mechanical design of axial flow	
	turbines	
17	- Rotor airfoil centrifugal stresses	
	- Rim web thickness	
18	- Disc of uniform stress	
_	- Disc thermal stresses	
	- Airfoil aspect ratio	
	Turbines technology	
19	- Materials	
19		
	- Manufacturing technology	
	- Balancing	
• •	Design of inlets	
20	- Subsonic inlets	
	- Supersonic inlets	
	Design of nozzles	
21	- Convergent nozzle	
	- Convergent-divergent nozzle	
22	- Thrust reversing and thrust vectoring	Dr. A. S. Al-Jaberi
	- Nozzle coefficients	
	- Nozzie coefficients	
	Inlets and nozzles technology	
23	- Materials	
	- Manufacturing technology	
	Accessory drives	
24	- Construction of gearboxes and	
	drives	
	- Engine power off takes	
	- Engine oil system	
	Engine control systems	
25	- Engine/Airframe interfaces	
23		
	- Control systems	
26	Engine starting	
26	- Fuel/Ignition control	
	- Engine rotation	
	- Throttle levers	
	- Starting sequence	
	Turbine engine inspection and	
27	maintenance	
28	- Inlet and compressor section	
29	- Combustion section	
30	- Turbine section	PID
	-Exhaust section	
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Subied	et Number: ANTE 438	
•	et: Aircraft Stability and Control	
Units:	•	
Weekl	y Hours: Theoretical: 2	
	Experimental: 1	
	Tutorial : 1	
Week	Contents	
	Introduction to A/C stability and	
1	control The freedom of motion of sineroft	
1	- The freedom of motion of aircraft - Basic axis	
2	Aircraft longitudinal static stability - Definition of longitudinal static	
4	stability and conditions	
	- General equation	
	Aircraft pitching moment	
3	- Pitching moment coefficient as	
-	a function of angle of attack	
l	- trimmed conditions	Dr. A. S. Al-Jaberi
	Applications	
4	- Examples	
	Directional static stability	
5	- General equations	
	- Conditions of stability	
	The longitudinal equations of motion-	
	A rigid symmetric A/C	
6	- Derivation of longitudinal	
	equations of motion	
7	Applications	
7	- Examples	
8	Longitudinal stability derivatives - All the longitudinal stability	
σ	derivatives equations	
	Equations of motion as a function of	
9	stability derivatives	
	- Derivation of longitudinal equations	
L	of motion in terms of the stability derivatives	
10	Applications	
10	- Examples	
	Longitudinal equations of motion	
11	solution Response to elevator as a function of	
11	- Response to elevator as a function of	ais 10
	A/C characteristic equation	
12	- Solution of characteristic equation	Here I
14	- Types of period oscillation	
	JPes of period obernation	المعادية الملاح

13	Derivation of lateral directional equations of motion and solution - Derivation of lateral directional equations of motion in terms of the stability derivatives - The solution	
14	Applications - Examples	
15	Yaw damper from stability and control augmentation - Principles of yaw damper - General equations - Effect of yaw damper on stability and control of A/C	
16	Introduction to non linear dynamics - General introduction - Stall dynamic - Wing rock	
17	Applications - Examples	Dr. A. S. Al-Jaberi
18	Gust equations derivation and its applications - Load factor estimation from the gust - Types of gust - Effect on response	
19	Longitudinal autopilot - Longitudinal equations of autopilot	
	Pitch orientation control system	
20	- Deriving basic equations of pitch orientation control system	
21	Flight path stabilization - Mach hold - Altitude hold	
	Vertical gyro as the basic attitude	
22	reference - Main idea of the gyro as the indicator of angle	
23	Lateral autopilot - Main assumptions	
23	Damping of Dutch roll	
24	- The principle of Dutch roll damping	
25	Yaw orientation control system	
25	- Derivation of the main equations Other lateral autopilot configuration	
26	·Introduction and principles	-140
27	Lateral transfer function for rudder displacement - Deriving the transfer function of control of the transfer function of	**
	basic rudder equation of motion	AS

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Dr. A. S. Al-Jaber

28	Lateral transfer function for ailerons - Deriving the transfer function of ailerons displacement
29	Lateral transient response for A/C - Explanation and the calculation methods of transient response
30	Applications - Examples



Subjec	t Number: ANTE 434	
•	t: Aircraft Structures	
Units:		
Weekly	y Hours: Theoretical: 2	
•••••••••••••••••••••••••••••••••••••••	Experimental: 1	
	Tutorial : 1	
Week	Contents	
	Introduction	
1	- General viewing	
	Structure components	
2	- Skin, Spar, Stiffeners and stringers	
	- Rib, Frame (Ring)	
	- Monocoque, Semi-monocoque	
	- Definitions	
	- Elastic axis, Shear center	
3	Aircraft construction materials	
3	- Wood - Aluminum alloys	
	- Aluminum alloys	
	- Steel alloys	Dr. A. S. Al-Jaberi
4	- Titanium alloys	
-	- Plastics	
	- Glass	
	- Composite materials	
	Wing cross section structure	
5	- Simple box beam	
	- Simple nose	
	- Center box	
	- Multi- flanges cell	
	- Multi-cells	
	- Multi-cells multi flanges	
	Bending of open and closed section	
6	beams	
U	- Introduction	
	- Sign conventions and notations	
7	- Resolution of bending moments	
	- Direct stress distribution due to bending	
	- Load intensity, shear force and bending moment relationships	
8	- Deflection due to bending	
9	- Approximations for thin-walled sections	
	General stress, strain and 80 00 0	
10	 displacement relationships For open and single cell closed section thin walled beams 	

11	Shear stress - Shear of thin-walled open tubes,	
10	without booms	
12	- Shear of thin-walled closed sections beams, without booms	
13	 Twist and warping of closed section beams Shear center 	
14	Pure torsion - Torsion of open section beams - Bredth-Batho formula	
	- Torsion of closed section beams	
-	Analysis of combined open and cl <mark>osed</mark>	
	sections	
15	- Bending	
	- Shear	
	- Torsion	
16	Structural idealization - Idealization of open and closed	
10	beam sections	
	Effect of idealization	Dr. A. S. Al-Jaberi
17	- Bending, shear and torsion of	
	open and closed section beams	
	Deflection	
18	- Deflection of open and closed	
	section beams	-
	Span-wise taper effect	
19	- Single web beam.	
19		
19	 Single web beam. Open and closed sections Beams having variable string areas 	
19 20	 Single web beam. Open and closed sections Beams having variable string areas Fuselage	
	 Single web beam. Open and closed sections Beams having variable string areas 	
	 Single web beam. Open and closed sections Beams having variable string areas Fuselage Bending 	
	 Single web beam. Open and closed sections Beams having variable string areas Fuselage Bending Shear Torsion Fuselage frame analysis 	
20	 Single web beam. Open and closed sections Beams having variable string areas Fuselage Bending Shear Torsion 	
20 21	 Single web beam. Open and closed sections Beams having variable string areas Fuselage Bending Shear Torsion Fuselage frame analysis Cut-outs in fuselage 	
20 21 22	 Single web beam. Open and closed sections Beams having variable string areas Fuselage Bending Shear Torsion Fuselage frame analysis Cut-outs in fuselage Wing 	
20 21 22 23	 Single web beam. Open and closed sections Beams having variable string areas Fuselage Bending Shear Torsion Fuselage frame analysis Cut-outs in fuselage Wing Bending 	
20 21 22	 Single web beam. Open and closed sections Beams having variable string areas Fuselage Bending Shear Torsion Fuselage frame analysis Cut-outs in fuselage Wing Bending Torsion 	
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20 21 22 23 24	 Single web beam. Open and closed sections Beams having variable string areas Fuselage Bending Shear Torsion Fuselage frame analysis Cut-outs in fuselage Wing Bending Torsion Shear Shear Shear center Taper effect Method of successive approximation/ Torsion Method of successive approximation/ Shear 	
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20 21 22 23 24 25 26 27	 Single web beam. Open and closed sections Beams having variable string areas Fuselage Bending Shear Torsion Fuselage frame analysis Cut-outs in fuselage Wing Bending Torsion Shear Shear Shear Shear Gut-outs in fuselage Wing Bending Torsion Shear Shear center Taper effect Method of successive approximation/ Torsion Method of successive approximation/ Shear Deflection Wing ribs analysis 	
20 21 22 23 24 25 26	 Single web beam. Open and closed sections Beams having variable string areas Fuselage Bending Shear Torsion Fuselage frame analysis Cut-outs in fuselage Wing Bending Torsion Shear Shear center Taper effect Method of successive approximation/ Torsion Method of successive approximation/ Shear Deflection 	
20 21 22 23 24 25 26 27	 Single web beam. Open and closed sections Beams having variable string areas Fuselage Bending Shear Torsion Fuselage frame analysis Cut-outs in fuselage Wing Bending Torsion Shear Shear Shear Shear Gut-outs in fuselage Wing Bending Torsion Shear Shear center Taper effect Method of successive approximation/ Torsion Method of successive approximation/ Shear Deflection Wing ribs analysis 	

الله اكبر

Dr. A. S. Al-Jaber

29	Fatigue - Safe life and fail-safe design
30	-Fatigue strength of components - Goodman method - Miner & palmgren method
	s-n curvePrediction of aircraft fatigue life



Subjec	t Number: ANTE 439]
•	t: Aircraft Systems and Maintenance	
Units:	•	
	y Hours: Theoretical: 2	
WUUM	Experimental: 2	
	Tutorial :	
Week	Contents	
	Aircraft systems	
1	- Introduction	
	- Basic definitions	
	Basic components used in aircraft mechanical systems	
	- Aircraft hardware	
2	- Pumps	
	- Valves and pipes	
	- Filters and cocks	
	Fuel system	
3	- Fuel system components	
4	- Fuel system operating modes	Dr. A. S. Al-Jaberi
	- Integrated civil aircraft systems	
5	- Troubleshooting	
	- Maintenance	
6		
6 7	Hydraulic system - Hydraulic system components	
/	- Types of hydraulic systems used in the modern aircrafts	
8	- Troubleshooting	
Ū	- Maintenance	
0	Pneumatic system	
9 10	- Pneumatic system components	
10	- Troubleshooting - Maintenance	
11	Environmental control systems	
	 The need for a controlled environment Environmental control system design 	
12	- Cooling systems	
	- Air distribution system	
13	- Air conditioning system	
	- Oxygen system	
	- cabin pressurization	
14	- De-ice systems	
17	- Troubleshooting	
	- Maintenance	No-THING
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	Emergency systems Al-AWSalt Decimical University	
15	- Warning systems	
	- Fire detection and suppression	

	- Emergency power sources		
	- Emergency oxygen		
	- Emergency systems maintenance		
	Flight control systems		
16	- Flight control surfaces		
	- Flight control linkage systems		
	- High lift control systems		
17	- Trim and feel		
	- Flight control actuation		
	- Fly-By-Wire control laws		
18	- Troubleshooting		
	- Maintenance		
	T 1		
19	Landing gear systems		
	- Construction		
	- Landing gear system maintenance	ا بد	
	- Inspection and maintenance of brake system and Tires		
	Airframe		
20	- Inspection		
	- Maintenance		
	Inspection fundamentals		Dr. A. S. Al-Jaberi
21	- General		
	- Required inspections		
22	- CHECKLIST		
23	- Aircraft Logs		
	- Special inspections		
24	- Publications Bulletins Maintenance manual		
	Overall manual		
	Structural repair manual		
25	Parts catalog		
	- Federal Aviation Regulation (FAR)		
	- Airworthiness directives		
	- Type certificate data sheets		
	Ground handling and support		
	equipments		
26	- General		
	- Starting engines		
	- Power units		
	- Air conditioning and heating units		
27	- Ground support air start units		
	- Preoiling equipment		
	- Aircraft fueling		
20	- Servicing aircraft with oil		
28	- Aircraft Tiedown		
	- Movement of aircraft		
	- Maintenance safety		
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	Al-Furat Al-Awsat Technical University		

29	CAA License - Qualifications structure
30	 EASA66/JAR66 syllabus modules and applicability Examinations and levels





Dr. A. S. Al-Jabe



•	t Number: CREQ 449 t: Computer Added Engineering	
Units:	• • •	
Weekl	y Hours: Theoretical:	
	Experimental: 3	
	Tutorial :	
Week	Contents	
WCCK	Introduction to CAE	
1	- Characteristics of CAE and its	
_	importance	
	Introduction to Modeling	
2	- Dealing with real physical objects	
	- Treating them as CAE models	
	Modeling of 2 Dimonsional Problems	
3	Modeling of 3-Dimensional Problems - Some examples of solid objects	
	of real problems	
	- Loads & boundary conditions	
4	- Utilization of symmetry to simplify problems modeling	
	e unzation of symmetry to simplify problems modeling	Dr. A. S. Al-Jaberi
_		
5	Reduction to Plane Problems	
(- Importance & conditions of	
6	reduction to plane problems	
7	- Reduction to axi- symmetrical models	
1	- Modeling 2-D plane stress, plane strain, and fluid flow	
	Bar, Beam, Problems	
8	- Analysis of bar, beam problems	
-	Plate Problems	
9	- Condition of reduction to plate model	
	- Load, Symmetry	
	Meshing; 2-D	
10	- Types of mesh	
	- Methods of meshing	
	Meshing 3-D	
11	- Types of mesh	
	- Methods of meshing	
	Meshing Bar, Beam, and Plate	
12	- Types of mesh	
	- Methods of meshing	
	Load – Structural	
13	- Types & implementation of	
	structural loads	NU-THUR
	Load – Thermal	
14	-Types & implementation of	11
	thermal loads	m / n
	and a man an amount of a more and	T/

	Load - Fluid	
15	- Types & implementation of	
10	fluid- flow loads	
	Solution- Structural	
16	- Static structural problems	
10	Solution – Dynamic	
17	- Dynamic structural problems	
	Solution – Thermal	
18	- Heat transfer problems	
	Solution- Fluid	
19	- Fluid flow problems	
	Results	
20	- Types of results	
	- Visual contours & data results	
21	- Results at internal sections	
	- Animation	
	Report Generation	
22	- Software- dependent generation of	
	eng. Reports of analyzed problems	
	Applications	
23	- Static 3D stress analysis problem	Dr. A. S. Al-Jaberi
24	- Steady- state heat transfer problem	
25	- Laminar flow problem	
26	- Modal analysis problem	
25	Importing Geometry	
27	Types & standards of solid geometryImporting from CAD systems	
	Contact Problems	
28	- Introduction to contact problems	
20	- Software- specific implementation	
	Plastic Deformation	
29	- Non-linear material behavior and	
<u> </u>	solution	
	Phase- change problems	
30	- Analysis & implementation of	
	phase- change problems	
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•	et Number: ANTE 418 et: Automatic Control	
Units:		
Weekl	y Hours: Theoretical: 2	
	Experimental: 1 Tutorial : 1	
Week	Contents	
1	Introduction and definitions - Basic definitions about the concepts of control	
2	Mechanical system and Transfer Function - Definition of transfer function - Deriving the transfer function for three basic parts of mechanical system	
3	Series and parallel connections in mechanical systems - Transfer function for mechanical system while connected it at series and parallel - Examples	Dr. A. S. Al-Jaheri
4	Torsional system - Deriving the transfer function for three basic parts of torsional System	
5	Electrical system, series and parallel connections - Deriving the transfer function for three basic parts of electrical system connected in parallel and series	
6	Thermal and fluid systems - Deriving the transfer function for thermal and fluid systems - Examples	
7	 Hydraulic system The basic concept of working the hydraulic system Deriving the transfer function of the system 	
8	Hydraulic servomotor system - Leverage system and deriving the transfer function for three cases of fixing - Method of connection with hydraulic system	*

9	 Pneumatic system The basic concept of working the pneumatic system Deriving the transfer function of the system 	
10	Block diagram - The principles of block diagram - The basic nine rules for reduction the block diagram	
11	Block diagram reduction - Method of reduction of block diagrams of multi-input and output	
12	Types of control and Laplace transformations - Types of control methods and basic functions of Laplace transformation	L
13	Test signals - The different types of test signals	
14	Response of first order system - Method of computing the response of first order system	Dr. A. S. Aklaberi
15	 Examples Response of second order system Method of computing the response of second order system Examples 	
16	Response specifications - The specification of response which determine the stability of system	
17	 Steady state error Computing the steady state error by using Toyler method and normal method and compare between them 	
18	Response improvement - The methods of response improvement - Examples	
19	System stability - The concept of system stability and its effect on control process	
20	Routh criterion - The Routh criterion for computing the stability of system	
21	Applications of Routh criterion Some applications about Routh criterion Examples 	
22	Root-locus method //// Technical University computing system stability	*

23	Rules of Root-locus method - Basic rules of root-locus	
	method - Examples	
	Polar-plot diagrams	
24	- The polar plot for computing system stability	
25	Principles of polar-plot diagrams - The method of polar plot diagram for	
	computing the gain - Examples	
	Logarithmic Scales and Bode Plots	
26	- Basic principles of logarithmic scale and Bode plots	
27	Construction of Bode Plots for Continuous-Time Systems - The method of construction of	
	Bode plots - Examples	
	Analysis of control system in state space	Dr. A. S. Al-Jaberi
28	- Principles and basic assumptions for state space method	
	State space representation of transfer	
29	function of systemThe state space representationExamples	
	Solving the time invariant state	
30	equations - The solution method of time invariant state equations	



•	et Number: ANTE 437	
Subjecture Units:	ct: Aircraft Vibration	
	y Hours: Theoretical: 2	
	Experimental: 2	
Week	Tutorial : Contents	
WEEK		
1	Introduction - Basic concepts of vibration and applications - Calculation of degree of freedom for systems	
2	Oscillatory motion - Introduction to oscillatory motion - Simple harmonic motion - Displacement velocity and acceleration relations	L
3	Free vibration of an undamped single degree of freedom system - Derivation of basic equation - Solving the eq. of motion and finding the natural frequency - Examples	Dr. A. S. Aldaberi
4	 Simple energy method (Raleigh principle) - conservative and non- conservative systems - Applying the simple energy method for different systems to find eq. of motion and natural frequency 	
5	Free vibration of viscous damped single degree of freedom system - Types of damping - Formulation and solving the equation of such system for different damping ratio - Examples	
6	Equivalent springs and dampers - Equivalent spring and damping for parallel and series connection - Examples	
7	 Logarithmic decrement Formulation of the basic eq. of logarithmic decrement Calculation the time for the decay of signal Examples 	
	Al-Furet Al-Aveet Technical University	**
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8	 Forced vibration of single degree of freedom system Formulation of the basic eq . of motion for damped and undamped forced vibration Behavior of the amplitude with w/wn (eq . of resonance) 	
9	Forced vibration for constant force - Behavior of the system with constant excitation force - Formulating the steady state and transient solution - Examples	
10	Forced vibration for sinusoidal force (Resonance conditions) - Behavior of the system with sinusoidal force - Formulation and solution of eq. - Equation for resonance	
11	- Examples Rotating unbalance - Explaining the unbalance forces Applications and examples	Dr. A. S. Al-Jaber
12	 Applications and examples Support motion Support motion and how this motion affect the motion of the system Examples 	
13	Vibration isolation - Transmissibility - Discussion of transmissibility behavior with w/wn for different damping ratios - Examples	
14	 Vibration measuring instrument The eq. of motion if the measuring device excited by a base force Amp., Vel., and acceleration that the device measure Examples 	
15 16	 Two degree of freedom system The equations of motion for 2- degree system Estimating the natural frequency and their mode shap Coordinate coupling Semi definite system Examples 	
	ATHIER ATTAXIER TECHNICAL UNIVERSITY	H J

 18 Lagrange equation Lagrange equation Lagrange ,eq. for damped & undamped system free and forced Vib . Applying it for several times Examples 19 Dynamic absorber (undamped) 	
- formulation of the eq. of dynamic absorber without damping - Examples	
20 Damped vibration absorber - formulation the eq. of dynamic absorber with damping - Examples	Dr. A. S. Al-Jaberi
21 Multiple degree of freedom system - Formulating the eq, of motion for multiple degree of freedom systems - Finding the natural freq and their mode shapes	
22 Influence coefficient matrix and stiffness matrix - Finding the eigen values and hence the natural frequencies and the eigen vector (mode shape) for multiple degree of freedom systems	
23 Wing moment of inertia - Calculating the wing equivalent moment of inertia - Examples	
 24 Torsional vibration Single degree of freedom systems Two degree of freedom systems Multiple degree of freedom systems (using holzer method) Torsional vibration for stepped shaft Torsional vibration for shaft with Gears 	

27	 Dunkerley's equation for aircraft wing system Estimating and application of 1st natural frequency for a system of external loads on aircraft wing 	
28	Rayliegh's equation for aircraft wing - Application of Rayliegh's equation to aircraft wing with lump masses, distributed and combined loads	
29	Application of iterative technique for aircraft wing and tail	
30	 Estimating the natural frequencies of aircraft wing and tail using the iterative technique from the lowest natural frequency and sweeping matrix Estimating the highest mode and natural frequencies using the iterative technique and stiffness matrix 	
		Dr. A. S. Al-Jaberi

